International Journal of Current Advanced Research

ISSN: O: 2319-6475, ISSN: P: 2319-6505, Impact Factor: 6.614

Available Online at www.journalijcar.org

Volume 7; Issue 3(I); March 2018; Page No. 11034-11038 DOI: http://dx.doi.org/10.24327/ijcar.2018.11038.1900



THE EFFECTS OF ARTIFICIAL AGING HEAT TREATMENT ON MECHANICAL PROPERTIES AND CORROSION BEHAVIOUR OF AA6082 ALUMINIUM ALLOYS

Adem Onat*

Sakarya University, Vocational School of Sakarya, 54290 Sakarya - Turkey

ARTICLE INFO

Article History:

Received 16th December, 2017 Received in revised form 20th January, 2018 Accepted 4th February, 2018 Published online 28th March, 2018

Key words:

AA6082 aluminium alloy, T6 heat treatment, Artificial aging, Mechanical properties, Corrosion behaviour

ABSTRACT

Aluminium alloys are now widely preferred in industrial applications due to their low density and high strength values. The properties of these alloys can be further improved by precipitation hardening heat treatment. For this purpose, the temperature and time used in the precipitation hardening process plays a very effective role on the performance of the material. Therefore, in this study, the effects of artificial aging heat treatment on mechanical properties and corrosion behaviour of AA6082 alloy were investigated. The effects of artificial aging time on microstructure and mechanical properties of alloy were analysed and at the same time corrosion behaviour was tried to be determined by corrosion tests. For experimental investigations, the alloy samples prepared in appropriate sizes were heated to 540°C (± 0.5°C) with heating rate 10°C/min. by electrical resistance ceramic furnace. For the solution heat treatment, samples were kept in the furnace at this temperature for 4 hours. The samples taken from the furnace were firstly poured into iced water at 10°C and then subjected to artificial aging at 190°C for 2, 4, 6, 10, 12 and 24 hours. Finally, the samples taken from the furnace were left to cool down in stagnant air. The results show that the mechanical properties of the material were increased with increasing aging time. Corrosion tests have shown that the corrosion resistance of the alloy depends on the artificial aging time. The best value of corrosion resistance was obtained at a temperature of 190°C at 10-hour aging period.

Copyright©2018 Adem Onat. This is an open access article distributed under the Creative Commons Attribution License, which permits unrestricted use, distribution, and reproduction in any medium, provided the original work is properly cited.

INTRODUCTION

Aluminium is the second most used metal in the world. Due to its high specific strength (strength/weight ratio), easy formability, high thermal conductivity, compatibility with surface treatments and resistance to corrosion, aluminium and its alloys are used in a wide range of applications from automotive, building and packaging sectors to high voltage-electricity transmission lines to construction applications [1-3].

6XXX series aluminium alloys, containing Mg and Si as main alloy elements have generally good extrusion and rolling capabilities. These alloys also have good corrosion resistance, especially in atmospheric environments. In addition to these favourable properties, the maintenance of the gloss of the anodized surface of the 6XXX series aluminium alloys also increases the amount of commercial use day-by-day [4-6]. At the same time, the strength of this alloy group can be substantially increased by a two-stage heat treatment [6,7].

This two-stage heat treatment consists of solutioning and ageing process.

*Corresponding author: Adem Onat Sakarya University, Vocational School of Sakarya, 54290 Sakarya – Turkey The solutioning process, holding the alloy in temperature below eutectic reaction, is used to dissolve the precipitates of Mg₂Si, homogenize the chemical elements concentration on the cross-section of dendrites of the α phase and the change in the silicon precipitations morphology. The solutioning process is usually performed at 460-540 $^{\circ}$ C for aluminium alloys of the 6XXX series.

The ageing process can be applied as natural aging at room temperature and the artificial aging at moderate temperatures. The ageing (soaking of the supersaturated alloy to separate strengthening phases from the supersaturated solid solution) the precipitation strengthening is obtained as a result of the phases precipitation of Mg₂Si, Al₂CuMg and Al₂Cu [8].

Artificial ageing (holding the alloy at constant temperatures during predetermined period) results in the improvement of the mechanical properties such as tensile strength and hardness and in simultaneous worsening of plasticity. Because of the growth of alloy's strength after heat treatment is very often accompanied with the reduction of the plasticity, their optimal relation should be selected depending on a given application of the alloy [8].

In the aging phase, which provides increased strength in the material, the precipitation process of the oversaturated melttakes place:

$$\alpha \to GP \to \beta^{II} (Mg_5Si_6) \to \beta^I (Mg_9Si_5) \to \beta (Mg_2Si)$$
 (1)

The maximum hardness is possible by obtaining the phase β^{II} during aging. On the other hand, the conversion of the β^{II} phase to the β phase is defined as over-aging, and the Face-Centred Cubic FCC crystal structure of the equilibrium phase β results in a reduction of the hardness of the alloy [9, 10].

Depending on the temperature and duration of the applied artificial aging heat treatment, the type, size and quantity of the precipitated phase play an active role on the mechanical properties of the aluminium alloy and corrosion behaviour [3, 6].

In this study, the effects of artificial aging on the hardness and corrosion resistance of AA6082 alloy were investigated.

Experimental Studies

The chemical analysis of commercial AA6082 aluminium alloy used in this study and in rod form is given in Table 1.

Table 1 The chemical analysis of AA6082 aluminium alloy (wt.%)

| Mg | Si | Mn | Fe | Cu | Cr | Zn | Ti | Al |
|------|------|------|-------|------|-------|-------|-------|-------|
| 0.66 | 0.96 | 0.49 | 0.204 | 0.02 | 0.004 | 0.014 | 0.016 | 97.63 |

The specimens, have dimensions of 12.09 mm in diameter and 15 mm height, were heated to $540 \pm 0.5^{\circ}$ C with heating rate 10° C/min. in an electrical ceramic furnace. They were subjected to a solutioning treatment process for 4 hours at this temperature. The samples taken from the furnace were first put into iced water at 10° C for 15mins.and then subjected to artificial aging at 190° C for 2, 4, 6, 10, 12 and 24 hours. Finally, the samples taken from the furnace were allowed to cool down in stagnant air.

The heat-treated samples were prepared metallographically to examine their microstructural properties. For this purpose, the samples passed through the emery papers of 400, 600, 800, 1000 and 1200 mesh respectively were polished using 3 μ m diamond paste solution and etched for about 30 seconds in Kellers Etch. Microstructure studies were performed using a JEOL JSM-6060LV brand SEM microscope.

The hardness measurements of the samples were carried out on a Brinell hardness tester using a 2.5 mm diameter hardened steel ball and a load of 62.5 kg. Hardness values were determined by taking the arithmetic average of at least 5 measurements for each sample.

Potentiodynamic polarization (Tafel) method was used to determine the corrosion behaviour of the samples. Three electrode techniques were used in the experiments performed in the Gamry potentiostat/galvanostat device, using saturated Ag/AgCl as the reference electrode and graphite electrodes as the auxiliary electrode. Experiments were performed at room temperature and 3.5% NaCl solution, and potentiodynamic polarization measurements were carried out at 5 mV/s for 3600 seconds with a voltage between -1 and 1. Prior to the corrosion tests, the surfaces of the specimens, 12.09 mm in diameter and 5.16 mm in dimensions, were passed through standard metallographic sample preparation steps up to 1200 mesh emery paper.

RESULTS AND DISCUSSION

The change in hardness values according to the aging time of samples subjected to artificial aging at different times at 190°C

after solutioning treatment for 4 hours at 540 $^{\circ}$ C is given in Figure 1.

The commercial hardness value of AA6082 aluminium alloy was 43 HB and after solutioning treatment the hardness reduced to 40 HB. As can be seen in Figure 1, the hardness value of the samples increases with increasing artificial ageing time. The highest hardness value (98 HB) was obtained at 10 h aging time at 190°C. The hardness tended to decrease again with the prolongation of the aging period and the hardness value fell to 89 HB in 12 hours and 78 HB in 24 hours respectively.

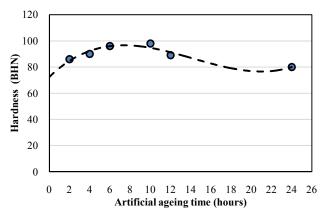


Figure 1 The effect of the artificial aging time on the harness value of AA6082 alloy

Based on the results of the performed investigations, a mathematical relationship between the hardness of the alloy and the heat treatment parameters can be established. Pezda, J. [8] informed the mathematical dependence of the effect of heat treatment parameters on the change of the alloy HB hardness as second order polynomial. The details of equation are given below:

$$HB = -1252,03 + 4,58X_1 + 101,1X_2 - 1,23X_2^2 + 0,94X_3 - 2,14X_4 - 0,4X_4^2 - 0,18X_1X_2 + 0,02X_1X_4 - 0,01X_2X_3 - 0,03X_2X_4 - 0,02X_3X_4$$
 (2)

 X_1 : solutioning temperature

 X_2 : solutioning time

 X_3 : ageing temperature

 X_4 : ageing time

At the end of the aging process, the increase in the hardness values of the alloys is due to the variations in the phases, precipitates and grain sizes formed in the microstructure. The increase in hardness is accepted as a sign of the success of the aging process after the dissolution in the literature [11]. The highest hardness value can be achieved by adjusting the optimum aging temperature and time.

When the aging process starts, both magnesium and silicon are beginning to precipitate as Mg_XSi_Y , as indicated by the formula of the solid solution. The precipitate phase formed at the start of aging is in perfect coherence with the aluminium matrix. But at high temperature, the Mg_XSi_Y precipitates become incompatible with the matrix together with increasing duration. As a result of the artificial aging process, the maximum hardness value of the samples is obtained by precipitation of β^{II} (Mg_5Si_6) phase in needle form [13]. The β^{II} phase begins to turn into a rod like β^I (Mg_9Si_5) phase during the increasing aging process, leading to a decrease in the hardness of the material [14]. As the aging time increases, the

precipitation of the β (Mg₂Si) phase becomes more dominant, resulting in a marked decrease in the hardness of the aluminum alloy [10]. In this case, full conformity is required to achieve higher strength and stiffness [12].

At the same time, coarse particles and constantly growing grain sizes are observed due to the combination of increasingly coexisting precipitates in the over aging period. In this period, the factors that prevent dislocation movements being decreasing and consequently the mechanical properties of the material are becoming increasingly smaller? [15].

Corrosion current (Icor) and corrosion potential (Ecor) values measured from the corrosion experiments of the samples by the potentiodynamic polarization (Tafel) method are given in Table 2.

Table 2. The corrosion test results of AA6082 samples

| Duration (Hour) | I _{cor} (μA) | E _{cor} (V) |
|--------------------|-----------------------|----------------------|
| 0 | 18.50 | -1.210 |
| 2 | 15.67 | -0.714 |
| 4 | 9.13 | -0.838 |
| 6 | 8.49 | -0.793 |
| 10 | 1.30 | -1.140 |
| 12 | 7.64 | -0.721 |
| 24 | 5.73 | -0.744 |

The following formulas were used to calculate the corrosion rate (Rcor) of the samples from these results [16]:

$$R_{cor} = \frac{I_{cor} \cdot K \cdot EA}{d \cdot A} \tag{3}$$

Where,

R_{cor} : Corrosion rate (mpy)

 I_{cor} : Corrosion Current Intensity (μA)

K : Constant

EA : Equivalent weight (atomic weight/valance)

d : Density (g/cm³) A : Surface area (cm²)

When the corrosion rate was calculated, the density of AA6082 alloy specimens exposed to T6 heat treatment, determined using the Archimedes technique, was 2.713 (g/cm³). The surface area of samples was 1,1499 cm² for corrosion test in this study and K constant was taken as 1.288 and EA (equivalent weight) was calculated according to the chemical composition of alloy the by formula given below [16].

$$EA = \frac{1}{\sum \frac{n.fi}{M_A}} \tag{4}$$

where;

n: Atomic valance of element f_i : Chemical composition of element M_{A} : Atomic weight of element

When the data of elements of AA6082 alloy are substituted in the formula, the equivalent weight value is found to be 9.0027. The corrosion rate (Rcor) was then calculated according to the formula given above. The change in corrosion rate with the heat treatment time is plotted in Figure 2.

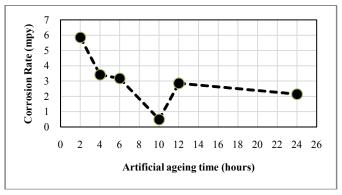


Figure 2 The change of corrosion rate with ageing time

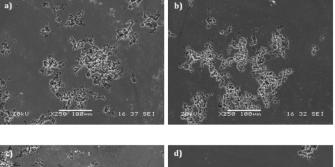
The corrosion rate of commercial alloy samples was detected as 6,91mpy. As can be seen from Figure 4, the corrosion rate rapidly decreases with aging time and obtained the lowest value of 0.49mpy in 10 hours, which is the optimum artificial aging time. In the case of over aging, the corrosion rate increases rapidly and reached to 2.85 mpy in 12 hours.

Aluminium and its alloys are highly resistant to corrosion due to the thin, protective barrier oxide layer on their surface. However, in the presence of chloride ions in the environment this protective oxide layer of the region is likely to suffer from corrosion. The most important factors affecting the corrosion resistance of aluminium alloys are the alloying elements they possess. However, temperatures and durations used in heat treatments also play a dominant role in corrosion behaviour [17].

After the corrosion test, surface morphologies of the samples were examined on a SEM microscope. A pronounced pitting corrosion was observed on the surfaces of each specimens (Figure 3). Similar results have been obtained in the literature reviews and it has been observed that with increasing aging period, the pitting corrosion also increases [18]. A similar situation has been detected by Yüksel [6] by the corrosion tests performed to AA6063 alloys artificial aged at different temperatures and durations. It has been found that intergranular corrosion decreases with increasing aging time, and with the increase of pitting corrosion and over-aging, pitting corrosion becomes dominant.

The corroded areas of the material surface were examined by SEM microscopy and the obtained SEM image and EDS analysis are given in Figure 4. The EDS analysis of the marked corrosion zone on the microstructure resulted in a concentration of chlorine ions (0.65%) in the corrosive solution, which is similar to many studies in the literature [16, 18]

When the aging process starts, both Mg and Si are beginning to precipitate in the form of solid solution Mg2Si and there is an Al matrix around this phase. In this case, electrochemically the Mg2Si phase is more active than the Al matrix so this situation causes the formation of a micro-galvanic corrosion cell and the intergranular corrosion between the phase bound to the grain boundaries and the matrix [6].



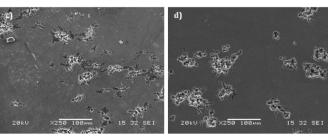
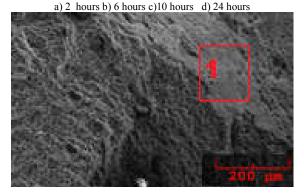


Figure 3 The variations of surface morphology of AA6082 alloy by ageing durations after corrosion tests



| The EDS An | The EDS Analysis of marked area | | | | |
|------------|---------------------------------|--|--|--|--|
| Element | Concentration | | | | |
| | (wt.%) | | | | |
| Si | 0.62 | | | | |
| Mg | 0.94 | | | | |
| Cĺ | 0.65 | | | | |
| Al | 98.44 | | | | |

Figure 4 Surface morphology after artificial aging AA6082 alloy after corrosion test and EDS analysis of corrosion products

As the selected temperature for artificial aging process increases, the aging time required to achieve optimum properties is shortened. In this case, however, the increasing aging time leads to the transition of the precipitated phase from needle-shaped to rod-like form [9]. The increase in pitting corrosion associated with increased aging time is due to the increase in the physical dimension of the more active precipitate phases compared to matrix. But this is not exactly proven [18, 19].

CONCLUSIONS

The obtained results from the artificial aging process applied to AA6082 aluminium alloy at 190°C for 2, 4, 6, 10, 12 and 24 hours and their pertinent discussion allow drawing the following conclusions:

- The AA6082 Al-Mg-Si alloy exhibits strong ageing response in spite of having very small amounts of Mg and Si
- 2. The experimental results have revealed that aging between 8 and 10 h at 190°C is the most suitable

- combination of time and temperature imparting maximum corrosion resistance and hardness to the alloy
- 3. Up to the completion of the precipitation of the β II phase, a significant increase in the hardness of the material with increasing aging time was obtained and the highest hardness value (98 HB) was detected at the 10-hour aging time.
- 4. With longer aging times, during the transformation of β^{II} phase to β^I and β phases (over-aging), the hardness of the alloy decreases rapidly as the sizes and distributions of the precipitates grow out of homogeneity and join with the adjacent precipitates and grow to extreme levels. During 24 hours of aging, the hardness value decreased to 78 HB.
- 5. Likewise, the corrosion rate of alloys is reduced with increasing aging time, the minimum corrosion rate (0.856 mpy) was obtained at 10 hour aging time
- 6. In the aging process of the AA6082 alloy, two types of corrosion were observed, namely intergranular corrosion and pitting corrosion.
- 7. During the artificial aging, the intergranular corrosion susceptibility of the alloy continues to decrease until the precipitation of the β^{II} phase is complete. However, as the size of the precipitated phase increased, the intergranular corrosion resistance increased, while the susceptibility of the alloy to pitting corrosion increased.

Acknowledgements

This study was supported by grants to the author from Sakarya University, Scientific Research Projects Coordination Unit project grant coded: 2016-50-01-040

References

- 1. Nandy, S., Bakkar, M. A., and Das, D. Influence of Ageing on Mechanical Properties of 6063 Al Alloy, Materials Today: Proceedings 2.4-5 (2015): 1234-1242.
- Mandava, S., Ramachandrula, S., &Yarramareddy, A. Effect of Thermal Treatment of a Ferro Magnetic Core on Induced EMF. Procedia Materials Science 6 (2014): 436-443.
- Wu, Y., Liao, H. Corrosion Behavior of Extruded near Eutectic Al–Si–Mg and 6063 Alloys. *Journal of Materials Science & Technology* 29.4 (2013): 380-386.
- 4. Panigrahi, S. K., Jayaganthan, R. A. Study on The Combined Treatment of Cryorolling, Short-Annealing and Aging for The Development of Ultrafine-Grained Al 6063 Alloy with Enhanced Strength and Ductility, Metallurgical And Materials Transactions A 41.10 (2010): 2675-2690.
- Vargel C. Corrosion of Aluminium, Elsevier, 2004, ISBN: 978-0-08-044495-4.
- 6. Yuksel B, Effect of Artificial Aging on Hardness and Intergranular Corrosion of 6063 Al Alloy, Pamukkale University *Journal of Engineering Sciences* 2017; 23. 4, (2017), 395-398
- Mohamed A.M.A. and Samuel F.H. "A Review on the Heat Treatment of Al-Si-Cu/Mg Casting Alloys", Heat Treatment - Conventional and Novel Applications, Dr. Frank Czerwinski (Ed.), InTech, (2012), DOI:

- 10.5772/50282.
- 8. Pezda, J. "The effect of the T6 heat treatment on hardness and microstructure of the EN AC-AlSi12CuNiMg alloy." *Metalurgija* 53.1 (2014): 63-66
- 9. Maisonnette D, Suery M, Nelias D, Chaudet P, Epicier T. Effect of Heat Treatments on The Microstructure and Mechanical Properties of A 6061 Aluminium Alloy, *Materials Science and Engineering* A, 528 (2011): 2718-2724.
- 10. Li H.Y., Zeng C.T., Han M.S., Liu J.J., Lu X.C. Time-Temperature-Property Curves for Quench Sensitivity of 6063 Aluminum Alloy, Transactions of Nonferrous Metals Society of China, 23. 1, (2013), 38-45.
- 11. Mrowka, GN, Sieniawski J, "Influence of Heat Treatment on the Microstructure and Mechanical Properties of 6005 and 6082 Aluminium Alloys, *Journal of Materials Processing Technology*, 162 (2005), 367-372.
- 12. Pratikno H., Aging Treatment to Increase the Erosion-Corrosion Resistance of AA6063 Alloys for Marine Application, Procedia Earth and Planetary Science, 14 (2015), 41-46.
- 13. Edwards G.A., Stiller K., Dunlop G. L., Couper M. J. The Precipitation Sequence in Al-Mg-Si *Alloys Acta Metallurgica* 46. 1, (1998), 3893-3904.

- El-Menshawy K, El-Sayed A.W.A, El-Bedawy M.E., Ahmed A.H., El-Raghy S.M. Effect of Aging Time at Low Aging Temperatures on the Corrosion of Aluminum Alloy 6061, *Corrosion Science*, 54 (2012), 167-173.
- 15. Meyveci A., Karacan, I., Caligulu, U., Durmus, H., "Pin-On-Disc Characterization of 2xxx and 6xxx Aluminium Alloys Aged by Precipitation Age Hardening" *Journal of Alloys and Compounds*, 491, 1–2, (2010), 278-283.
- 16. Baboian R. NACE Corrosion Engineer's Reference Book (4th Edition). NACE International, 2016.
- 17. Jones D. Principles and Prevention of Corrosion Second Edition, Prentice-Hall, 1996.
- Svenningsen G, Larsen M.H., Nordlien J.H., Nisancioglu K. Effect of High Temperature Heat Treatment on Intergranular Corrosion of AlMgSi (Cu) Model Alloy, Corrosion Science, 48.1, (2006), 258-272.
- Svenningsen G, Larsen MH, Walmsley JC, Nordlien JH, Nisancioglu K. Effect of Artificial Aging on Intergranular Corrosion of Extruded AlMgSi Alloy with Small Cu Content. *Corrosion Science*, 48. 6, (2006), 1528-1543.

How to cite this article:

Adem Onat (2018) 'The Effects of Artificial Aging Heat Treatment on Mechanical Properties and Corrosion Behaviour of Aa6082 Aluminium Alloys', *International Journal of Current Advanced Research*, 07(3), pp. 11034-11038. DOI: http://dx.doi.org/10.24327/ijcar.2018.11038.1900
